

Applications of the SSME Real-Time Vibration Advanced Engine Health Management **Monitoring System**

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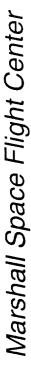
What is the Real-Time Vibration Monitoring System (RTVMS)?

- RTVMS is a 32-channel high speed vibration data acquisition and processing system developed at Marshall Space Flight Center (MSFC).
- Delivers sample rates as high as 51,200 samples/second per channel
- Performs Fast Fourier Transform (FFT) processing via onboard digital signal processing (DSP) chips in a real-time



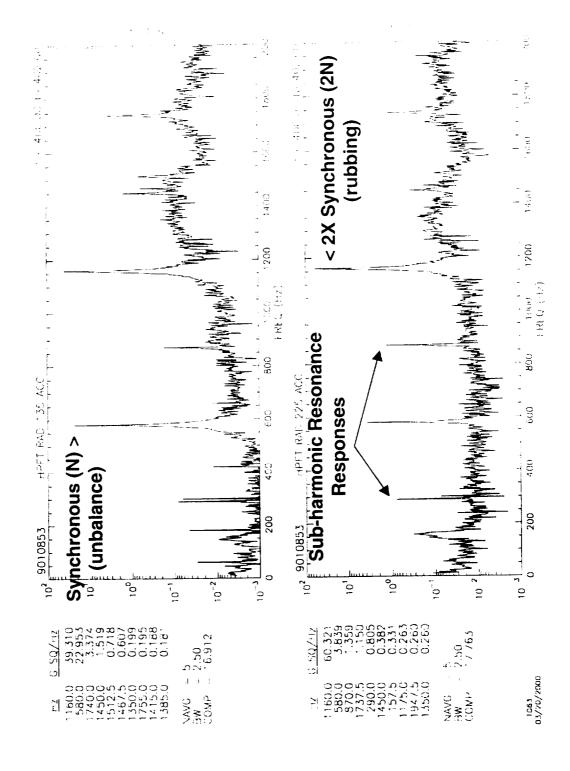
Why is RTVMS important?

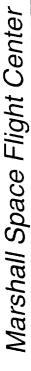
- Real-time FFT processing yields real-time vibration spectral data.
- Advanced engine health assessment is achieved by utilizing the vibration spectra to provide:
- accurate sensor validation
- enhanced engine vibration redlines
- indicators of imminent failure can be assessed and utilized to mitigate catastrophic engine failures - a first in rocket engine Discrete spectral signatures (such as synchronous) that are health assessment.
- High sample rates provide for enhanced time and frequency resolution over a broader frequency spectra.





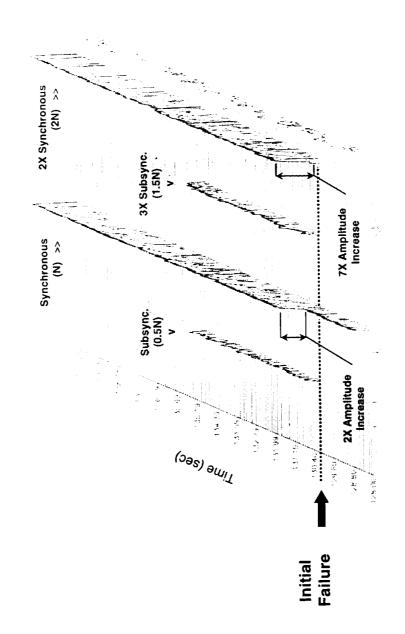
- Vibration spectral signatures utilized by the RTVMS







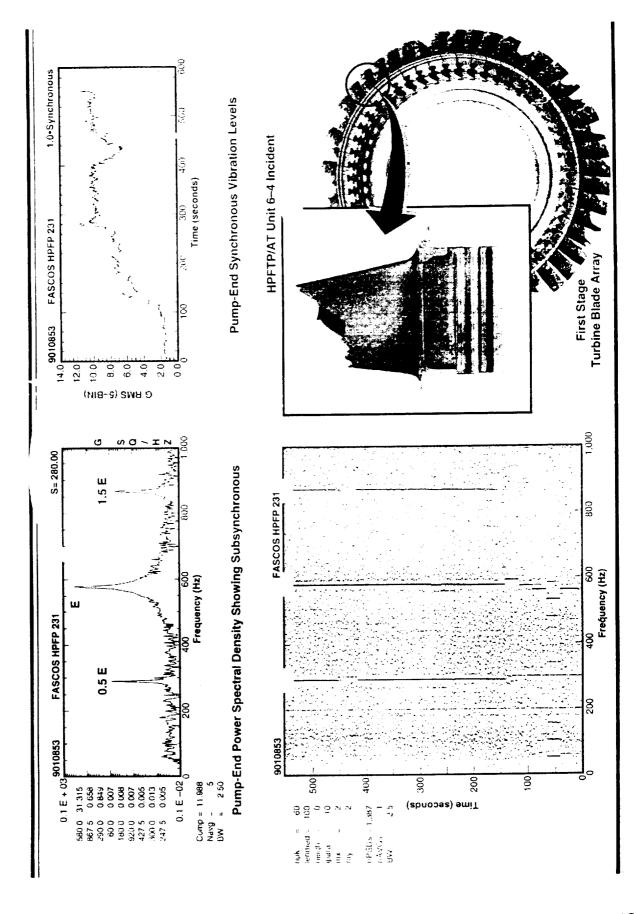
- RTVMS spectra yields both frequency and amplitude trends -



Frequency (Hz)

RTVMS Data Surrounding 130 Second Event during test 901-853 - Time 128-140 seconds

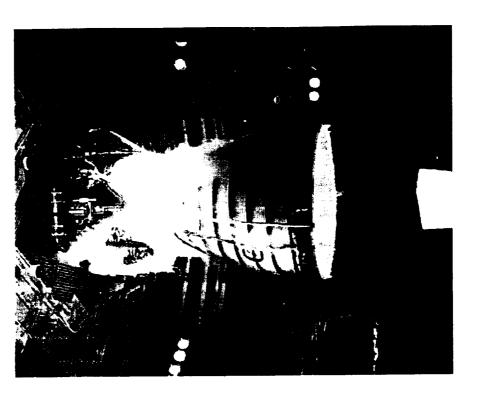






Operational History - Ground Testing

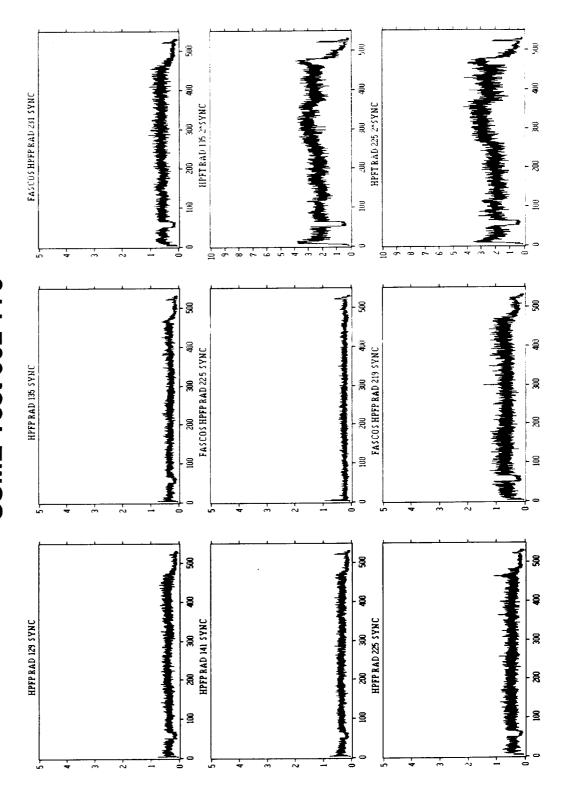
- RTVMS has been deployed at the Stennis Space Center since October 1996
- RTVMS has actively monitored over 150 SSME static hot-fires.
- The system monitors 3 separate engine vibration redlines
- 2 high pressure fuel turbopump (HPFTP) vibration redlines
- 1 high-pressure oxygen turbopump (HPOTP) vibration redline.
- The RTVMS installed at SSC provides:
- data acquisition at 20,480 samples/second for 32 channels
- real-time vibration redline amplitude trackings and power spectral densities (PSD's)
- automatic engine test termination in 100 milliseconds (50 millisecond resolution)







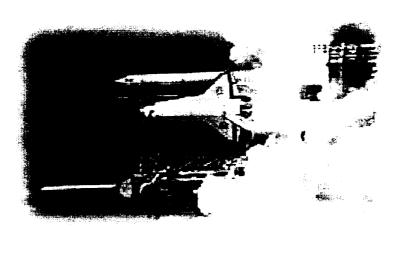
RTVMS Real-Time Tracking Display **SSME Test 902-770**





Operational History - Flight

- A sub-scale version of RTVMS flew aboard
 STS-96 as part of the HTD-2 flight experiment.
- During the flight, the RTVMS module:
- acquired data from 8 vibration measurements (one flight engine) at 10,240 samples/second
- processed the digital data real-time
- actively located and monitored the synchronous vibration responses for the HPOTP and HPFTP for flight duration
- provided real-time discrete frequency and amplitude trackings of both high pressure turbopumps
- The RTVMS flight experiment proved the concept of high-speed vibration data acquisition and realtime processing in a flight environment.



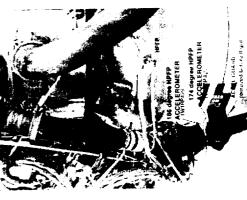


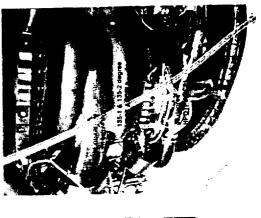
HTD-2 SSME RTVMS

- Components
- (1) Ruggedized high-speed data acquisition (A/D) board
- (1) Ruggedized digital signal processing (DSP) board
- MSFC turbopump signature tracking algorithm
- (8) Existing SSME vibration measurements consisting of:
- (3 ea) High-Pressure Fuel Turbopump (HPFTP) and High-Pressure Oxidizer Turbopump (HPOTP) accelerometers
- » (1 ea) Gimbal Bearing and Oxidizer Preburner (OPB) accelerometers



- Engine vibration health monitoring
- Benefit
- Mitigation of engine catastrophic failures
- Real-time high-speed digital acquisition and processing
- Reduced post-flight processing
- Range
- 266 G peak-to-peak
- Data Availability
- T-10 seconds to MECO + 15 seconds

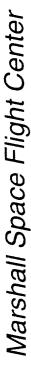






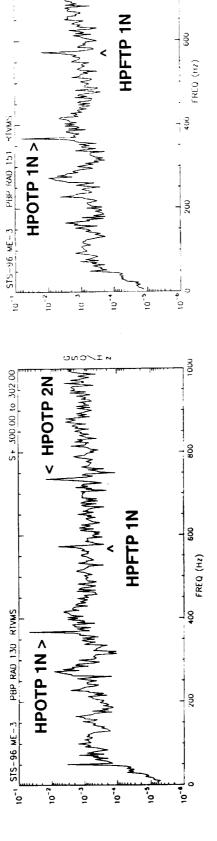
HTD-2 SSME RTVMS Results

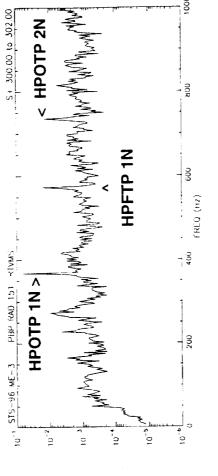
- Successful acquisition of accelerometer data at 10,240 samples/second per channel
- Data was written to on-board flash storage.
- All digitally acquired data was accurately processed, real-time, by the DSP board.
- Produced real-time frequency spectra .
- Discrete frequency responses were available for in-flight monitoring and analysis.
- The MSFC algorithm examined the frequency spectra real-time during engine operation for the synchronous frequency response.
- Synchronous is the primary indicator of SSME turbopump rotordynamic health.
- The algorithm accurately located synchronous for both high-pressure pumps
- Synchronous was monitored real-time throughout engine operation.
- All synchronous frequency and amplitude tracking results were written to onboard flash storage.

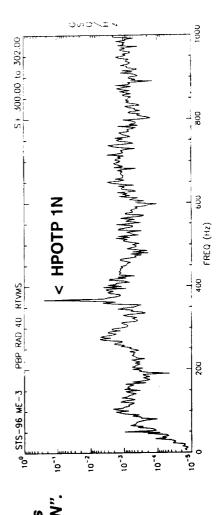




RTVMS HPOTP Power Spectral Densities STS-96 SSME Position 3





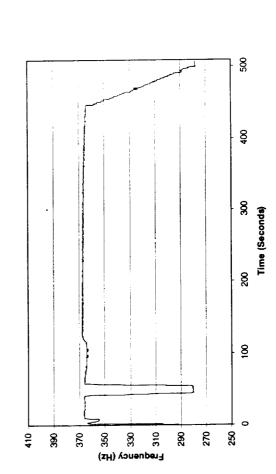


response is labeled as "1N". ** Note: The synchronous

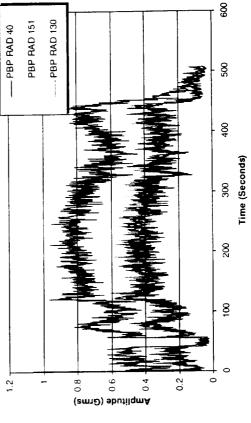


HTD-2 RTVMS HPOTP Tracking Results STS-96 SSME Position 3

RTVMS HPOTP Synchronous (Speed) Frequency Tracking



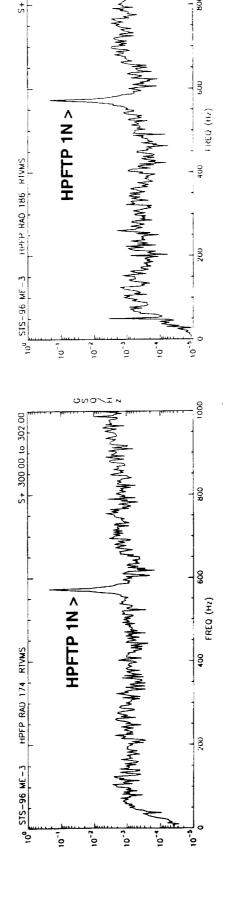
RTVMS HPOTP Synchronous Amplitude Trackings

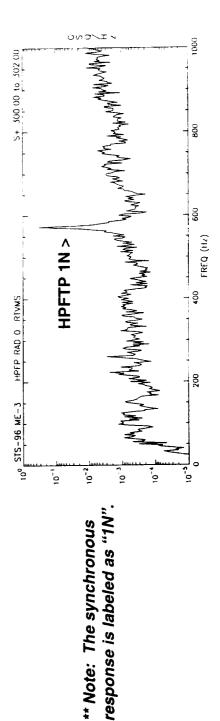






HTD-2 RTVMS HPFTP Tracking Results STS-96 SSME Position 3





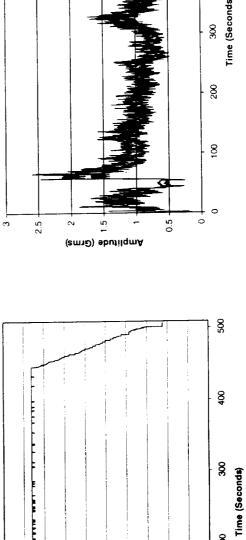




HTD-2 RTVMS HPFTP Tracking Results STS-96 SSME Position 3

RTVMS HPFTP Synchronous (Speed) Frequency Tracking

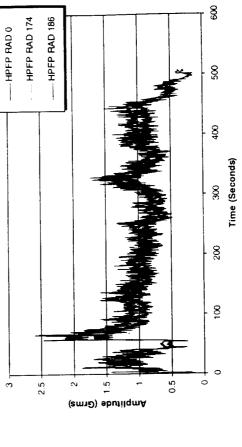
RTVMS HPFTP Synchronous Amplitude Trackings



Frequency (Hz)

6

0



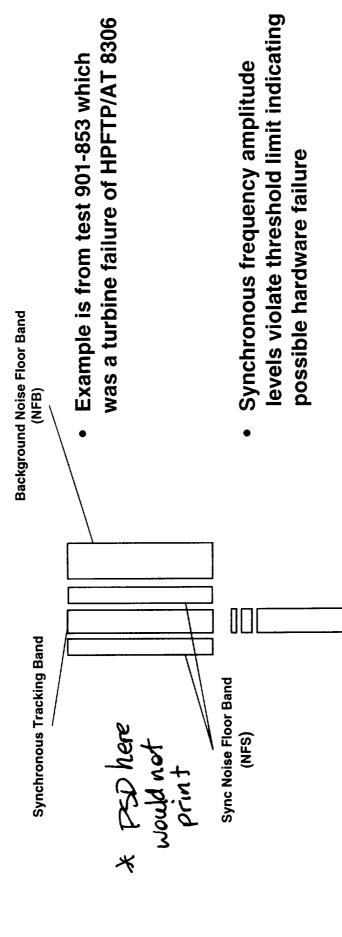


RTVMS and the Advanced Health Management System (AHMS)

- RTVMS is the basis for the SSME AHMS Shuttle Safety Upgrade Program
- AHMS Phase I is a modified SSME Controller which will incorporate the RTVMS synchronous vibration redline methodology.
- AHMS Phase II is the Health Management Computer (HMC) which will incorporate the full RTVMS analysis package module.
- MSFC/TD63 has also developed sensor validation software which will reside on RTVMS on AHMS Phase I and II



AHMS RTVMS Synchronous Redline Logic Example



Sensor validation logic reviews factors and determines sensors are valid

20 18 16 12 10

Amplitude (Grms)

4



009

500

200

100

Time (Seconds)



HMC RTVMS Advanced Analysis Methodology

- real-time contain numerous spectral responses pertinent to pump health The six turbopump accelerometers that will be processed and analyzed
- Analysis of the frequency spectra can be performed out to 10,000 Hz
- Current analysis plans for the HMC RTVMS include:
- tracking and redline monitoring of synchronous vibration response (N)
- primary indicator of pump health (mass unbalance indicator)
- tracking and analysis of synchronous harmonics
- 2N and 3N (primary indicators of internal rotor rubbing)
- 4N/8N (HPOTP) and 6N (HPFTP) (blade wake responses from pump impellers)
- as bearing deadband interaction) and limit-cycle whirl (rotor instability) detection of sub-harmonic resonance (forced vibration response such responses



- **HMC RTVMS Advanced Analysis Methodology**
- Future analysis upgrades will include:
- determination and tracking of bearing related frequencies (Cage, Ball Spin, Inner Race and Outer Race) to determine the health of the bearings
- cavitation detection and active signature phase correlation algorithms
- rotating and non-rotating related turbopump phenomena nonlinear algorithms to distinguish differences between
- active unknown anomaly identification and monitoring



Conclusions

- potential catastrophic SSME turbomachinery failures through RTVMS delivers the capability to detect and mitigate real-time extraction of discrete vibration frequency components.
- The AHMS HMC with the RTVMS, Linear Engine Model (LEM), and Optical Plume Anomaly Detector (OPAD) subsystems provides an advanced, reliable health management capability for the SSME.